

Lecture II a. - Transport Heuristics

a) \rightarrow Transport \leftrightarrow Mixing \rightarrow Profile

- Pipe Flow
- Stellar Convection

b) \rightarrow Scalings - MFE

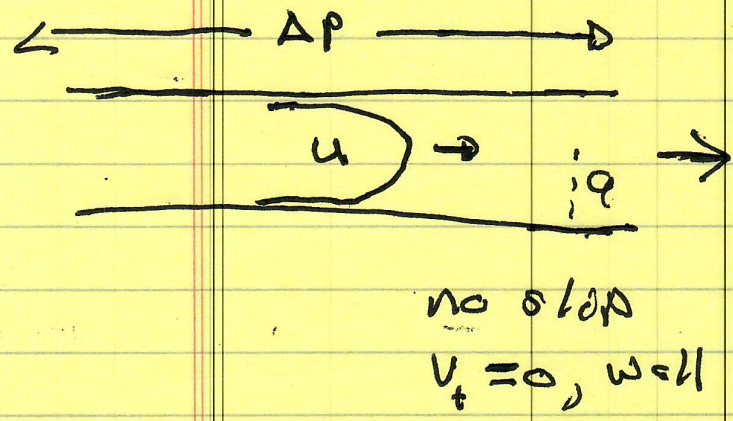
- Flux, D
- ρ_* - the critical ratio
- $\Lambda \rightarrow$ Gyro-Bohm, Bohm and between
- Shearing Effects

→ Transport ↔ Mixing ⇒ Profile

a.) Pipe Flow - Turbulent
(Navier-Stokes)

(cf. Landau & Lifshitz)

→ inhomogeneous, bounded system



$\frac{\Delta P}{l}$ → drive,
pressure drop
per length.

keeping score:

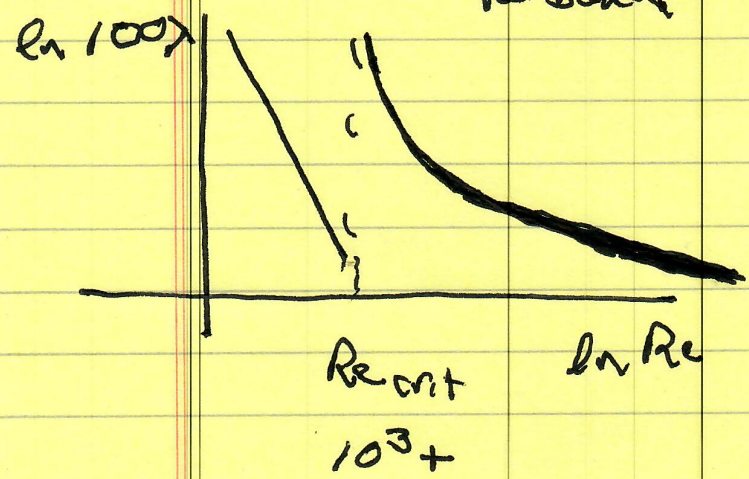
$$\left(\partial_x v + v \cdot \partial v = -\frac{\Delta P}{\rho} + \nu \partial^2 v \right)$$

$$\lambda = 2 \alpha \Delta P / l / \frac{1}{2} \rho U^2$$

→ resistance factor

(akin $T_E \sim W / P_{in}$)
laminar. turbulent

(shear-driven)



$$Re = \frac{2 U \alpha}{\nu}$$

better $1/\lambda$ vs
 Re
or K_F vs. ΔP



more rapidly than in turbulent flow.

Figure 32 shows a logarithmic graph of λ as a function of R . The steep straight line corresponds to laminar flow (formula (43.6)), and the less steep curve (which is almost a straight line also) to turbulent flow. The transition from the first line to the second occurs, as the Reynolds number increases, at the point where the flow becomes turbulent; this may occur for various Reynolds numbers, depending on the actual conditions (the intensity of the perturbations). The resistance coefficient increases abruptly at the transition point.

$$\lambda \sim \frac{\Delta p}{\rho u^2 / 2}$$

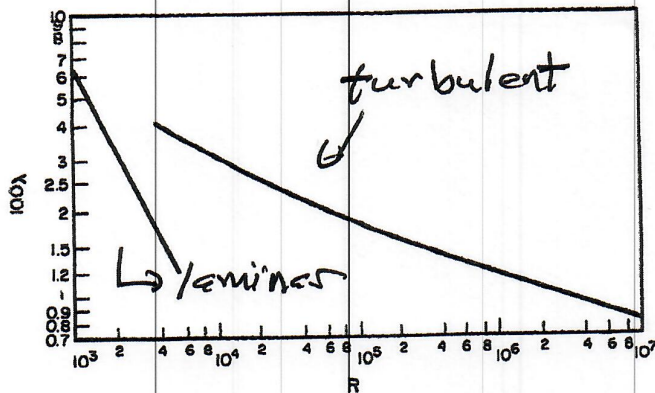


FIG. 32

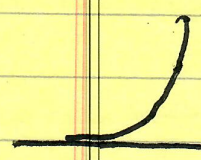
Re

→ turbulent resistance curve ⇒

Momentum confinement scaling.

→ also → universal boundary layer structure

core → flat/ply



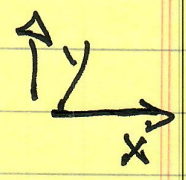
→ log law ↔ empirical

Mandel 25, 132.

"universal"

→ describe different flows of different size, etc. Re (so long as turbulent)

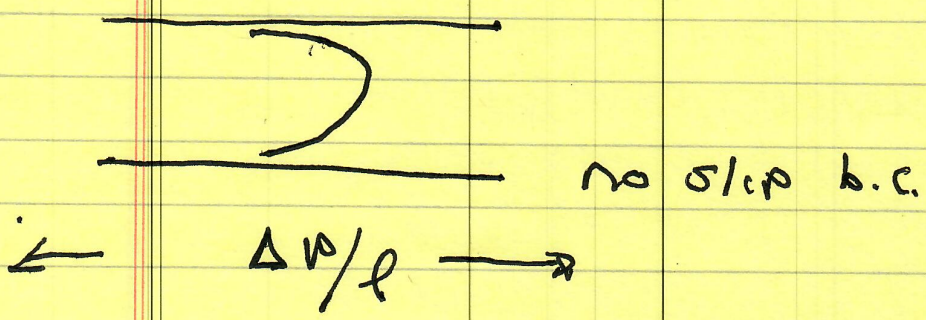
→ same shape $U(y)$ profile



↔ counterpart of h^4 , etc.

⇒ counterpart of "profile consistency", "resiliency", "stiffness" etc.

What is going on here?



- drag on flow \rightarrow $\lambda \rightarrow$ due momentum flux to wall
- turbulent transport/mixing of momentum

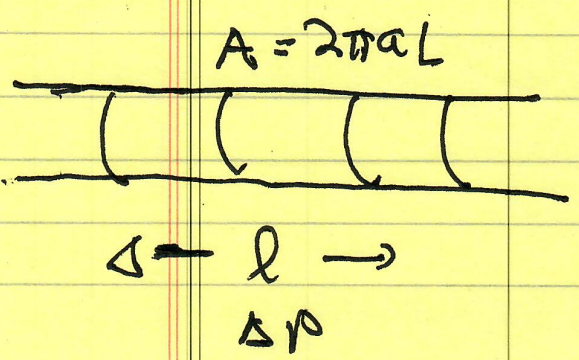
\Rightarrow

- wall stress must balance pressure drop.

so

$$\tau_{wall} = \rho U_*^2$$

$U_* \equiv$ typical turbulent velocity



Force on Wall \sim

$$\rho U_*^2 A_{wall}$$

$$\sim \rho U_*^2 2\pi a l$$

and Force on Fluid (per l) \sim (Pressure drop) \times A
 $\sim \Delta P \pi a^2$

so balance \Rightarrow

$$\rho U_*^2 (2\pi a l) = \Delta P (\pi a^2)$$

$$U_* = \left(\frac{\Delta P}{2\rho} \right)^{1/2} \left(\frac{a}{l} \right)^{1/2}$$

- \sim "Prandtl's velocity" \sim characteristic velocity
- \sim "typical velocity" of turbulence in (turbulent) pipe flow

N.B. \sim viscous vs turbulent stress? \rightarrow
thd

$\sim U_* \sim$ isotropic $\langle v_i v_j \rangle \rightarrow U_*^2$
 \downarrow
 Reynolds stress.

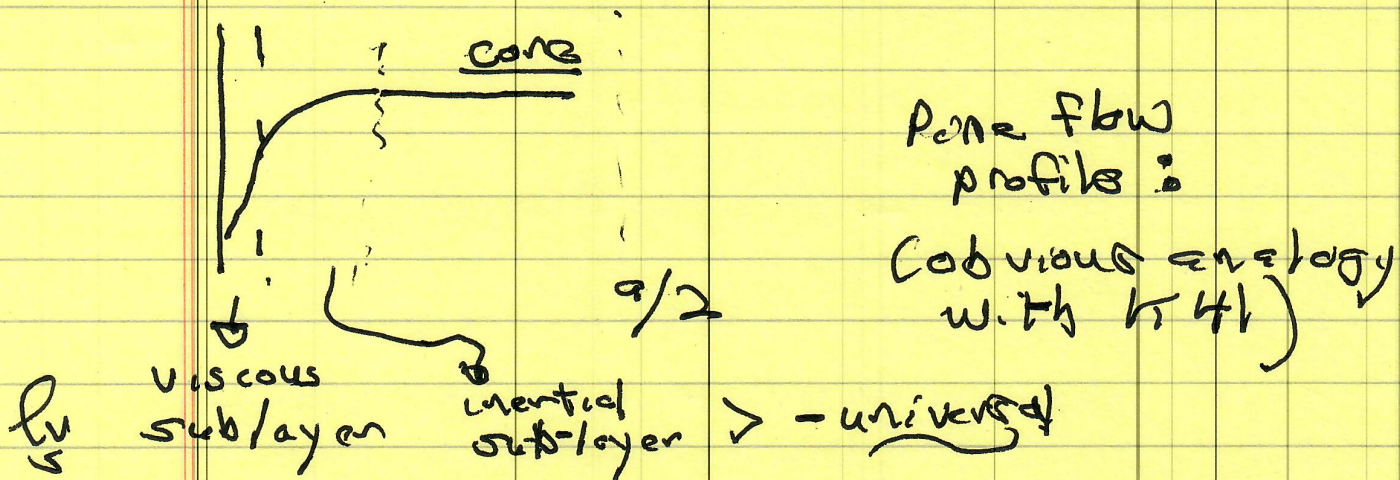
\sim Laminar $\left[\begin{array}{l} - \text{Poiseuille} \\ - \text{parabolic} \end{array} \right]$
 $-r \nabla^2 v_x = -\frac{2x\rho}{\rho}$

→ THE Question

- What is the profile $u(y)$?

~ practical question - skin MFE

→ Dimensional Reasoning



Can note:

Length scales: $a \rightarrow$ radius

$$l_{vs} \sim \frac{\nu}{u_*} \rightarrow \text{viscous sublayer scale}$$

in small scale region, close to wall

$$\tau_{yx} = \rho U_*^2 = -\rho \frac{\partial U}{\partial y}$$

viscous transport

$$U(y) = \frac{\rho U_*^2}{\rho} y$$

$$= \frac{U_*^2}{\rho} y$$

→ Linear profile
viscous sublayer

y ≡ distance from wall.

But what of inertial sublayer?

$$\delta_{vis} < y < \delta$$

Only parameters: U_* , ρ , μ

y
↓
distance from wall

Key: inertial sublayer is scale invariant

So, if seek velocity gradient

$$\frac{\partial U}{\partial y} = ?$$

Only possibilities are U_*
 $y \rightarrow$ distance from wall

$$\frac{\partial U}{\partial y} = \frac{U_*}{y}$$

Log Profile

$$\Rightarrow U = U_* \ln(y)$$

$$\rightarrow = U_* \ln(y/l_w) + U_*$$

(measured from $y=0$)

to match to viscous sublayer at l_w .
 Constant K (Von Karman) enters.

- Heuristic Reasoning

Consider turbulent mixing as a momentum transport process, akin to kinetic theory of gases.

Flux driven transport ↓

$$\tau_w = \rho U_*^2 = \rho \langle \tilde{u}_y \tilde{u}_x \rangle$$



transport via distribution of slugs/parcels of momentum.
 \rightarrow conserved locally.

$$\overline{v_x^2}(y) = \overline{u_x^2}(y - l) - u_x^2(y)$$

↓
parcel
scattered

$$\approx \text{[blacked out]} - l \frac{\partial u_x}{\partial y}$$

$$\overline{v_y^2} \sim u_x^2$$

$$\tau_w = -\rho \langle u_x l \rangle \frac{\partial u_x}{\partial y}$$

momentum diffusivity } "eddy/turbulent viscosity"

What is l → mixing length (analogous to μ)

Scale of variance ⇒

Mixing length restricted only by distance to nearest boundary (i.e. no scale)

So \rightarrow mixing length restricted only
by distance to (nearest) wall

$$\therefore l \sim y$$

$$\tau_w \approx -\rho U_* y \frac{\partial U_x}{\partial y}$$

and
$$-\rho U_* y \frac{\partial U_x}{\partial y} = \rho U_*^2$$

$$U_x = U_* \ln y + C$$

(with const.)

(measured
from $y=0$)

$$U_x \rightarrow \frac{U_*}{K} \ln(y/l_w)$$

inertial Layer
profile \rightarrow
"Law of the
Wall"

Profile \rightarrow turbulent pipe flow

\Rightarrow Welcome to Prandtl \rightarrow the "mother" of
all mixing length theory.

Some comments:

→ as in K41, clear phenomenology critical to model.

- data!

→ many ongoing:

- studies of transport, turbulence physics:

streaks, vortices, staircases, anomalous scaling, staircase - ...

but

→ Log law works pretty well!
(can fit $\ln u$ vs $\ln Re$ plot)

→ why single U_* ?

→ value of turbulent velocity?

→ From mixing of mean & gradient

$$\bar{u} \sim l \frac{\partial u}{\partial y} \sim l \frac{U_*}{y} \sim \cancel{y} \frac{U_*}{\cancel{y}}$$

→ Scale separation?

- Better Question!

ie for diffusive model, should have

$$l_{mix} / L_{macro} < 1 \rightarrow \text{c.f. validity Chapman Enskog.}$$

here $l_{mix} \sim \gamma$

$$L_{macro}^{-1} \sim \frac{1}{u} \frac{\partial u}{\partial y}$$

$$\sim \frac{1}{u + (l_{mix} \gamma)} \cdot \frac{u \gamma}{\gamma}$$

$$\text{so } l_{mix} / L_{macro} \sim \frac{1}{l_{mix} \gamma} \leq 1$$

→ Marginal ..

- note γ/γ scaling

but

- it works!

c.f. Spiegel (posted) 63
for more.

→ What of turbulent dissipation?

Consider N-S Eqn:

$$\frac{\partial \underline{u}}{\partial t} + \underline{\tilde{u}} \cdot \nabla \underline{\tilde{u}} + \langle v_x \rangle \frac{\partial \underline{\tilde{u}}}{\partial x} + \tilde{u}_y \frac{\partial \langle v_x \rangle}{\partial y} = -\nabla p + \nu \nabla^2 \underline{\tilde{u}}$$

$\underline{\tilde{u}}$ and avg:

$$\nabla \cdot \underline{u} = 0$$

$$\frac{\partial \langle \tilde{u}^2 \rangle}{\partial t} + \langle \cancel{\underline{\tilde{u}} \cdot \nabla \frac{\tilde{u}^2}{2}} \rangle + \langle v_x \rangle \langle \cancel{\underline{\tilde{u}} \cdot \frac{\partial \tilde{u}}{\partial x}} \rangle + \langle v_y v_x \rangle \frac{\partial \langle v_x \rangle}{\partial y} = - \langle \cancel{\underline{u} \cdot \nabla p} \rangle - \nu \langle (\nabla \tilde{u})^2 \rangle$$

odd
 odd
 odd
 odd

so

$$\frac{\partial \mathcal{E}}{\partial t} = - \underbrace{\nu \langle (\nabla \tilde{u})^2 \rangle}_{\text{viscous dissipation}} - \underbrace{\langle \tilde{u}_y \tilde{u}_x \rangle \frac{\partial \langle v_x \rangle}{\partial y}}_{\text{Reynolds work}}$$

Reynolds work
 (input of energy to ~~the~~ fluctuations from mean flow)

obviously > 0 .

define:

$$\epsilon = \langle \tilde{v}_y \tilde{v}_x \rangle \frac{\partial U_x}{\partial y}$$

↓

turbulent dissipation rate

Can use MLT:

$$\langle \tilde{v}_y \tilde{v}_x \rangle = U_x \times \frac{\partial U}{\partial y}$$

∴

$$\epsilon = (U_x \times) \frac{\partial U}{\partial y} = \nu_T \left(\frac{\partial U}{\partial y} \right)^2$$

↓
 "rate of heating" of
 fluctus by turb.
 viscous relaxation

ultimately:

$$\int_V \langle (\tilde{v})^2 \rangle = \int \nu_T \left(\frac{\partial U_x}{\partial y} \right)^2 \quad \left[\begin{array}{l} \text{in steady} \\ \text{state.} \end{array} \right]$$

$$\begin{aligned} \epsilon &= (U_{xy}) \left(\frac{U_x}{y} \right)^2 \\ &= U_x^3 / y \end{aligned}$$

dissipation rate
largest near
wall $(\propto U_x)$

ϵ finite as
 $v \rightarrow 0$

Proof ? - Pipe counterpart of 4/5
Law ?!

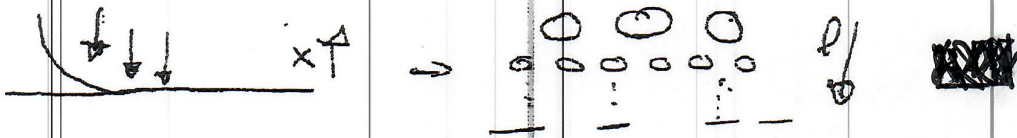
N.B. Mixing Length Theory, while not
rigorous, is

- useful \rightarrow especially in
complicated problems.

- non-trivial

\downarrow
Staircase,
Flowover ice

c.f. Spiegel '63
(p. 24)



→ Now, interesting to tabulate comparison between Pipe Flow and K41 Problem

Pipe Flow (Prandtl)	K41 (Kolmogorov)
scales: $a, x, \nu/u_*$	l_0, l_n, l_d
invariance: $x \rightarrow$ real space	$l \rightarrow$ scale space
inertial sublayer	inertial range
viscous sublayer	dissipation range
balance: $u_*^2 = \nu_T \frac{\partial u}{\partial x}$	$\epsilon = \frac{\nu(l\epsilon)^2}{l^3}$
physics: eddy viscosity $\nu_T = u_* x$	turn-over rate $1/\tau(l) = \frac{\nu(l)}{l}$
velocity: $u = \frac{u_*}{R} f(x)$	$\nu(l) = \epsilon^{1/3} l^{2/3}$
universal profile	universal spectral scaling
invariance: $\nu = \nu_T$ $x_0 = \nu/u_*$	$\nu(l)/\epsilon = \nu/l^2$ $l_d = \nu^{3/4} / \epsilon^{1/4}$